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SHUTTLE CRITICAL ITEMS LIST - ORBITTE	MUMBER: 06-132-0538-03						
SUBSYSTEM: ARS - COOLING	REVISION: 11/07/88						
LEG : PRI COOL PUMP AND ACCUM ITEM MAKE: VALVE, WATER BYPASS CONTROL	CRITICALITY OF THIS YAILURE MODE: 1R2						
FAILURE MODE: EXTERNAL LEARAGE							
MISSION PRASE:							
LO LIFT-OFF CO ON-ORBIT DO DE-ORBIT							
	COLUMBIA DISCOVERY ATLANTIS						
CAUSE: MECHANICAL SHOCK, VIBRATION, CORROSION, DEGRADATION	MATERIAL DEFECT, SEAL MATERIX						
CRITICALITY 1/1 DURING ANY MISSION PHASE OF ABOUT? N							
REDUNDANCY SCREEN A) PASS							
B) PASS							
C) PASS							
A)							
B)							
c)							
- FAILURE STREETS -							

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SEUTTLE CRITICAL ITEMS LIST - ORBITER MUMBER: 06-182-0528-03

(A) SUBSTSTEX:

LOSS OF REDUNDANCY - LOSS OF ONE WATER COOLANT LOOP.

(B) INTERFACING SUBSTSTEX(8):

LOSS OF COOLING OF AFFECTED WATER COOLANT LOOP. FREE WATER IN CARIN.

(C) MISSION:

POSSIBLE EARLY MISSION TERMINATION DUE TO LOSS OF ONE WATER COOLART LOOP.

(D) CREW, VERICLE, LEG ELEMENT (8) : POTENTIAL LOSS OF CREW/VEHICLE UPON SUBSEQUENT LOSS OF REDUNDANT WATER COOLANT LOOP.

RATIONALE FOR CRITICALITY:

- DISPOSITION RATIONALE -

(A) DESIGN:

ACTUATOR OPERATED TWO-WAY DIVERTER VALVE WITH A CYLINDRICAL VALVE ELEMENT. HAS REDUNDANT DYNAMIC SEALS USED TO PREVENT WATER LEAKAGE TO AMBIENT. VALVE MATERIAL IS CORROSION AND MEAT RESISTANT STEEL SAE 30347 CONDITION A AND SEALS ARE VITON.

(B) TEST:

ACCEPTANCE TEST - PROOF PRESSURE TESTED AT 135 PSID. EXTERNAL LEARAGE TEST - 0.01 CC/MR WATER MAX AT 90 PSIG.

QUALIFICATION TEST - BURST PRESSURE 180 PSID. COLLAPSE PRESSURE 22.5 PSID. PROOF PRESSURE 135 PSID. LEAKAGE - 0.01 CC/HR WATER MAX AT 90 PSIG. VIEBATION, SHOCK AND ACCELERATION DONE WITH THE PUMP PACKAGE: 1168 OPEN/CLOSE CYCLES VS 500 CYCLES FOR MISSION REQUIREMENT. SUBJECTED TO RANDOM VIBRATION SPECTRUM ENVELOPE OF 20 TO 150 HZ INCREASING AT 6 DE/OCTAVE TO 0.03 G**2/HZ, CONSTANT AT 0.03 G**2/HZ FROM 150 TO 1000 HZ, DECREASING AT 6 DB/OCTAVE FROM 1000 TO 2000 HZ FOR 48 MINUTES PER AXIS IN THREE ORTHOGONAL AXES. DESIGN SHOCK - THREE TERMINAL SANTOOTH FULSES OF 20 G PEAK AMPLITUDE AND 11 MS DURATION APPLIED IN BOTH DIRECTIONS ALONG EACH OF THREE ORTHOGONAL AXES.

IN-VEHICLE TESTING - SYSTEM DECAY TEST IS PERFORMED AT 85 - 95 PSIG, 8 CC/MIN MAX LEAKAGE. FUMP OUT PRESSURE AND ACCUMULATOR QUANTITY ARE CONTINUOUSLY MONITORED WHEN THE VEHICLE IS POWERED UP AND SERVE AS AN

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INDICATION OF EXTERNAL LEAKAGE.

OMRSD - FUMP ACCUMULATOR QUANTITY AND OUTLET PRESSURE ARE CONTINUOUSLY MONITORED WHILE THE VEHICLE IS POWERED UP DURING EACH TURNAROUND, AND SERVE AS AN INDICATION OF EXTERNAL LEAKAGE. WATER IS SAMPLED PER SPEC SE-5-0073 DURING SERVICING.

(C) IMSPECTION:

RECEIVING INSPECTION

INCOMING PARTS ARE VERIFIED FOR MATERIALS AND PROCESS CERTIFICATION. RECEIVING LOG IS MAINTAINED FOR VERIFICATION.

CUNTAMINATION CONTROL

CLEANLINESS REQUIREMENTS VERIFIED BY INSPECTION. CORROSION PROTECTION IS VERIFIED BY INSPECTION. CORROSION PROTECTION PROVISIONS ARE CHECKED.

ASSEMBLY/INSTALLATION

ORIENTATION OF COMMECTOR MASTER KEYWAY IS DETERMINED AND VERIFIED BY INSPECTION. TORQUE APPLIED TO HOUNTING SCREWS IS CHECKED PER DRAWING REQUIREMENT. BOTH COMPONENTS AND ASSEMBLY ARE INSPECTED VISUALLY. SUPER KOROPON TREATED SURFACE OVERCOATED WITH POLYURPTHANE IS VERIFIED BY INSPECTION.

NONDESTRUCTIVE EVALUATION LEAK TEST IS VERIFIED BY INSPECTION.

CRITICAL PROCESSES

WELD OF THE FILTER BOUSING TO PUMP INLET FLANGES IS VERIFIED. TUBE WELD IS CHECKED TO VERIFY ITS INTEGRITY DURING ASSEMBLY. COMPONENT HEAT TREATMENT IS VERIFIED BY INSPECTION.

TESTING

ATP IS VERIFIED BY INSPECTION.

HANDLING/PACKAGING

PACKAGING FOR SHIPMENT IS VERIFIED BY INSPECTION.

(D) FAILURE ELSTORY:

NO FAILURE HISTORY APPLICABLE TO EXTERNAL LEAKAGE FAILURE MODE. THE CONTROL VALVE HAS SUCCESSFULLY PERFORMED WITHOUT FAILURE THROUGH THE DURATION OF THE SHUTTLE PROGRAM.

(E)	OPERATIONAL	VAE:
TBS.		

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SHUTTLE CRITICAL ITEMS LIST - ORBITTE

MUMBER: 06-182-0536-03

RELIABILITY ENGINEERING: N. L. STEISSLINGER:
DESIGN ENGINEERING : N. K. DUONG U.
QUALITY ENGINEERING : D. R. STOICA ON :
NASA RELIABILITY :

NASA DESIGN

HASA QUALITY ASSURANCE :